

Department of Transport

Supplemental Type Certificate

This approval is issued to:

AMT Helicopters Pty. Ltd.

5/19 Lear Jet Drive

Caboolture, Queensland 4510

Australia

Number: SH18-66

Issue No.: 1

Approval Date: N

November 22, 2018

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Responsible Office:

Prairie and Northern

Aircraft/Engine Type or Model:

Bell 206A, 206B

Canadian Type Certificate or Equivalent:

H-92

Description of Type Design Change:

Installation of Tail Rotor Blades IAW CASA STC SVR 520.

Installation/Operating Data, Required Equipment and Limitations:

Installation and Maintenance of the Composite Tail Rotor Blade must be in accordance with Civil Aviation Safety Authority of Australia (CASA) approved AMT Helicopters Pty Ltd Installation and Maintenance Manual CTRB-IMM, Revision 6 dated June 14th, 2017 or later CASA approved revision.

Compliance with the Airworthiness Limitation Section approved by CASA in AMT Helicopters Pty Ltd Installation and Maintenance Manual CTRB-IMM, Revision 6 dated June 14th, 2017 or later CASA approved revision is mandatory.

Operation of the Composite Tail Rotor Blade must be in accordance with CASA approved AMT Helicopters Pty Ltd Flight Manual Supplement Doc. No. 209/373/FMS1, Revision IR, dated October 20th, 2014 or later CASA approved revision.

Definition of the Type Design Change: CASA approved Master Records Index AMT Helicopters Pty Ltd Doc No. 209/373/MRI, Revision 8 dated June 14th, 2017 or later CASA approved revision.

Basis of Certification for Affected Areas:

CAR 6 effective 20 December 1956: 6.111, 6.112, 6.113, 6.114, 6.115, 6.120, 6.121, 6.122, 6.123, 6.130, 6.140, 6.225. And FAR 27 effective 1 February 1965 including amendments up to FAR 27-44: 27.1, 27.177, 27.301, 27.303, 27.305, 27.307, 27.391, 27.395, 27.397, 27.571, 27.601, 27.602, 27.603, 27.605, 27.607, 27.609, 27.610, 27.611, 27.613, 27.619, 27.623, 27.625, 27.629, 27.653, 27.659, 27.661, 27.671, 27.1301, 27.1309, 27.1529, 27.1565, 27.1581. Plus 14 CFR Part 27.573 at Amdt 27-47.



Conditions: This approval is only applicable to the type/model of aeronautical product specified therein. Prior to incorporating this modification, the installer shall establish that the interrelationship between this change and any other modification(s) incorporated **will not** adversely affect the airworthiness of the modified product.



Loy Y. Liew For Minister of Transport

